

Stored Mechanical Energy and Carbon Onion Resilience Under Space Radiation

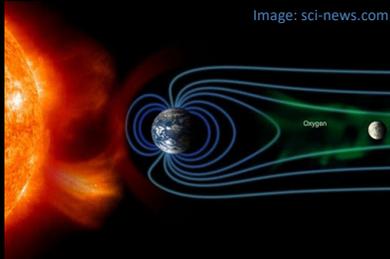
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INTRODUCTION

The lunar environment is extreme due to: vacuum, thermal stress and radiation. Nano-carbons are a promising approach to lubrication challenges in extreme environments including vacuum, radiation and non-terrestrial temperature regimes.

Image: sci-news.com



METHODS

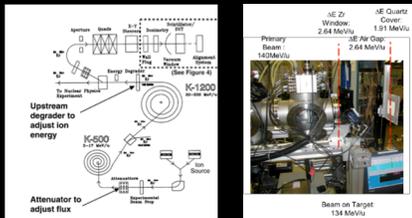
Researchers at Tokyo Institute of Technology have investigated the vacuum tribological performance of major types of nano-carbons [1a, b]. The best vacuum tribological performance was achieved with carbon onions (COs).

Carbon Lubricant Performances in Air and Vacuum [1 a, b]

Friction Coefficient	Carbon Onions	CNTs	C ₆₀	Graphite	VGCs
Air	0.1-0.2	0.5	0.5	0.2	0.5
Vacuum	0.04-0.06	0.2	0.6	0.6	0.2

This research further indicated that nano carbons have a high tolerance for thermal stress [1 a].

The present work focuses on the study of radiation effects on COs using well-calibrated heavy ion beams at realistic solar wind energies at the National Superconducting Cyclotron Laboratory (now FRIB) [2, 3].

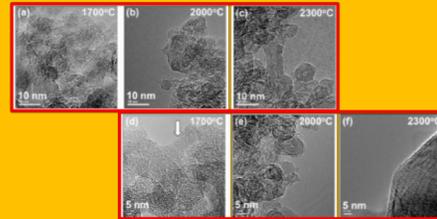


RESULTS

EXPERIMENTS:

- Relativistic heavy ion interactions with COs synthesized at 1700 °C, 2000 °C and 2300 °C, were investigated using primary beams of ~90% fully stripped argon-40 at 140 MeV/u, and calcium-48 at 140 and 70 MeV/u.
- Irradiation times that resulted in 10,000 Gray (Joule/kg) cumulative total dose (TD) for each sample proved sufficient to introduce non-destructive structural changes.
- Pre and post radiation characterization of temperature-series COs by HRTEM was performed in a JEOL 2200FS operated at 200 kV.

PRE-RAD: Tokyo Institute of Technology COs synthesized at 1700 °C and 2300 °C displayed well-known increasing polygonal character as a function of increasing growth temperature.



POST-RAD: The 1700 °C COs showed an increase in the polygonal character and fusion of adjacent onion pairs (arrow).

The 2000 °C COs showed fewer layer breaks in individual graphene shells and an overall increase in polygonal character that caused them to resemble pre-rad 2300 °C COs.

The 2300 °C COs showed the greatest variety of rearrangements, with the development of large crystallites of planar graphite (shown), long, 5-7 layer planar graphene ribbons, and occasional amorphous regions.

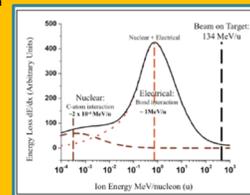
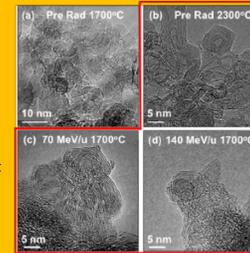
ENERGY VERSUS TIME:

Calcium-48 experiments were also performed with 140 versus 70 MeV/u to investigate energy versus time equivalence for the same TD

POST-RAD: Post irradiation 1700 °C COs at (c) 70 MeV/u and (d) 140 MeV/u calcium-48 both resemble 2300 °C pre-irradiation carbon onions: **energy-time TD equivalence in the CO system.**

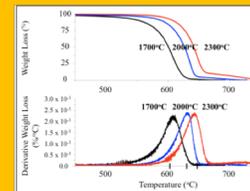
TOTAL ENERGY LOSS: Monte Carlo SRIM calculations for total energy loss dE/dx with individual nuclear and electronic interaction contributions [2]:

- Peak of the **nuclear contribution** at 2×10^{-4} MeV/u far from the beam-on-target 134 MeV/u energy.
- Peak total energy loss \leftrightarrow **peak electronic contribution** at 1 MeV/u but still far from the beam-on-target 134 MeV/u energy.

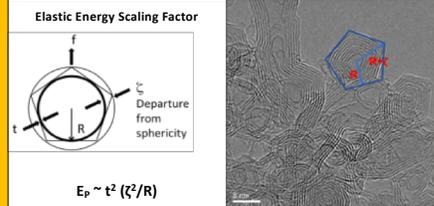


THERMOGRAVIMETRIC ANALYSIS (TGA):

- Nonlinear increase in ease of **dissociation** with increasing polygonal character.
- Most evident in **derivative weight loss**



ANALYSIS

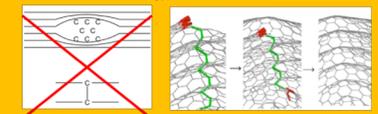


Variable	1700 Pre-Rad	2000 Pre-Rad	2300 Pre-Rad	Ar 1700 Post-Rad	Ar 2000 Post-Rad	Ar 2300 Post-Rad	Ca 1700 Post-Rad	Ca 1700 Post-Rad
Specimen Rad	0.706	0.942	1.40	1.156	1.591	1.708	1.141	1.403
Avg Z (nm)	/0.155	/0.125	/0.116	/0.221	/0.214	/0.214	/0.145	/0.138
Avg # of Layers	6.4	5.7	6.3	6.3	7.8	7.2	6.7	7.2
Thickness	/0.371	/0.3	/0.3	/0.213	/0.952	/0.412	/0.559	/0.52
Avg. Layer Thickness (nm)	0.346	0.453	0.463	0.433	0.379	0.349	0.435	0.441
Thickness	/0.329	/0.027	/0.033	/0.031	/0.018	/0.017	/0.024	/0.027
Avg. Radius (nm)	2.502	2.923	3.093	2.643	3.432	3.141	2.787	3.057
t ² (°C/R)	0.024	0.062	0.136	0.095	0.106	0.113	0.088	0.125

ANALYSIS applies Ref [4] approach to heavy ion irradiation of COs
RESULT: Polygonal increase \leftrightarrow stored elastic energy increase as shown by experimental scaling factors
PRE-RAD: Temperature-induced polygonal increase
POST-RAD: Radiation-induced polygonal increase

CONCLUSIONS

Graphene layer rearrangement driven by low-energy dislocation migration mechanisms only available in multi-layer **radial** situations [5] is consistent with Total Energy Loss results.



Highly polygonal COs are typically considered less defective/more perfect. However, our results indicate that release of the **increased elastic energy stored in highly polygonal COs can drive increased dislocation-driven rearrangement(s).**
Lower temperature synthesis COs may be more resilient to heavy ion induced rearrangement(s) and therefore preferable for lunar environments. A practical conclusion is that the polygonal character of a CO sample should be quantified prior to space lubricant applications.

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